BELL MODEL 206B JET RANGER II FLIGHT MANUAL SUPPLEMENT FOR LIGHTWEIGHT EMERGENCY FLOTATION

LANDING GEAR

206-706-211

APPROVED JUNE 26, 1984

This supplement shall be attached to the Flight Manual when the 206–706–211 Lightweight Emergency Flotation Landing Gear has been installed.

The information contained herein supplements the information of the basic Flight Manual. For Limitations, Procedures, and Performance Data not contained in this supplement, consult the basic Flight Manual.



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FAA APPROVED 206B FLIGHT MANUAL

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NOTE: Revised text is indicated by a black vertical line. Insert latest revision pages; dispose of superseded pages.					

INTRODUCTION

The Bell High–Skid Gear with Lightweight Emergency Floats, Kit No. 206–706–211, consists of a high–skid landing gear, emergency floats attached to the main skid panels, inflation system, navigation lights, and attaching hardware. Installation of this kit permits operation over land or water. Float inflation time is approximately 5 seconds.



TYPE OF OPERATION

Operation with the pop-out floats inflated is limited to a flight to a servicing facility for repacking and recharging the system. Amphibious operations are not approved.

The floats and covers must be installed and ground handling wheels removed for all flight operations.

Accomplish preflight float system check daily prior to performing over water operations.

Flight operations requiring use of the external hoist are PROHIBITED and the system SHALL BE DEACTIVATED when the floats are installed unless Cable Guard kit 206-706-214 is installed.

If Cable Guard Kit is installed, hoist cable and hook shall be stowed prior to float inflation.

AIRSPEED LIMITATIONS

FLOATS STOWED

Floats stowed, covers installed – Same as basic helicopter.

Doors ON or OFF in any combination — Same as basic helicopter.

FLOATS INFLATED

Maximum inflation airspeed -60 mph(52 knots) IAS.

AIRSPEED LIMITATIONS (Cont)



During the inflation cycle undesirable pitching will occur at airspeeds above 60 mph (52 knots) IAS).

Maximum allowable airspeed, floats inflated — 80 mph (69 knots) IAS.

Maximum AUTOROTATION airspeed, floats inflated — 70 mph (60 knots) IAS.

RATE OF CLIMB LIMITATIONS

Maximum rate of climb with floats inflated is 1000 feet per minute.

CENTER OF GRAVITY LIMITS

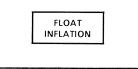
Actual weight changes shall be determined after kit is installed and ballast readjusted, if necessary, to return empty weight CG to within allowable limits. Refer to Center of Gravity vs Weight Empty Chart in Maintenance Manual.



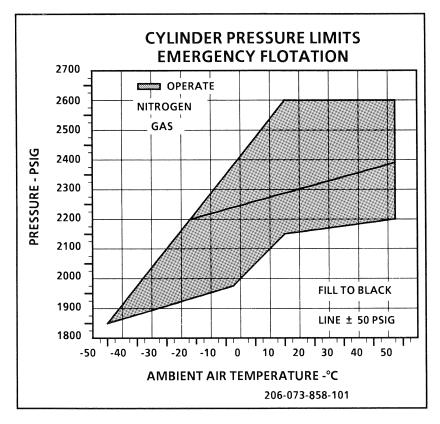
FLOAT ARMING/INFLATION ABOVE 60 MPH PROHIBITED

MAXIMUM ALLOWABLE AIRSPEED FLOATS INFLATED - 80 MPH (69 KNOTS) IAS

MAXIMUM AUTOROTATION AIRSPEED, FLOATS INFLATED -70 MPH (60 KNOTS) IAS



HOIST CABLE AND HOOK MUST BE STOWED PRIOR TO FLOAT INFLATION



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NORMAL PROCEDURES

EXTERIOR CHECK

Floats stowed.

Nitrogen lines — Check for condition and evidence of leakage.

Float covers clean and secured.

Float inflation cylinder — Check for proper temperature and altitude vs inflation pressure. Refer to placard on cylinder. Check cannon plug for security.

INTERIOR CHECK

PREFLIGHT FLOAT SYSTEM CHECK

 $\label{eq:FLOATMANUALARM} FLOATMANUALARM switch - OFF (guard closed).$

FLOAT POWER circuit breaker - Check IN.

FLOAT TEST and ARMED lights — PUSH TO TEST. (If installed.)

CAUTION LIGHT PANEL — Actuate test switch. (If installed.)

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PREFLIGHT FLOAT SYSTEM CHECK (Cont)

FLOAT TEST switch - FLOAT TEST position, and hold.

FLOAT INFLATION trigger switch — PULL ON, float test light ON, then release.

FLOAT TEST switch – RELEASE, test light OUT.

FLOAT MANUAL ARM switch — POWER (guard open), FLOAT ARMED light — Check ON, then switch OFF (guard closed), light OFF.

IN-FLIGHT OPERATIONS

OVER WATER OPERATION

FLOAT MANUAL ARM switch – POWER (guard open).

FLOAT ARMED light — Check ON.

During flight at altitudes above 500 feet and at airspeeds of 60 mph (52 knots) IAS and above the system SHOULD BE DEACTIVATED by positioning the FLOAT MANUAL ARM switch in the OFF position (GUARD CLOSED).

Rearm system prior to landing.

OVER LAND OPERATION

FLOAT MANUAL ARM switch – OFF.

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DESCENT AND LANDING - FLOATS STOWED



If the CG is aft of station 112.7, practice of autorotational touchdowns should be avoided due to nosedown pitching.



Run–on landings on other than a hard, firm surface should be exercised with caution.

NOTE

Tail-low, run-on landings should be avoided to prevent nosedown pitching.

EMERGENCY PROCEDURES

FLOAT INFLATION PROCEDURE

Reduce airspeed below maximum inflation airspeed -60 mph (52 knots) IAS.

Establish autorotation or low power descent at approximately 500 feet per minute.

FLOAT MANUAL ARM switch - POWER (guard open).

FLOAT ARMED light – Check ON.

FLOAT INFLATION trigger switch - PULL ON.



DO NOT inflate floats more than 2000 feet above anticipated landing surface.

NOTE

During flight with floats inflated, a random bumping of the skid gear crosstube against the landing gear saddles will occur. Reducing airspeed will reduce bumping.

AFTER EMERGENCY WATER LANDING

After landing, inspect the helicopter for possible damage. If malfunction was cause of landing, correct malfunction.

If no damage has occurred to helicopter and malfunction has been corrected, the helicopter can be ferried to the nearest maintenance facility to repack floats and charge system. The ferrying airspeed is restricted to 80 mph (69 knots) IAS. The maximum rate of climb while ferrying is 1000 feet per minute.

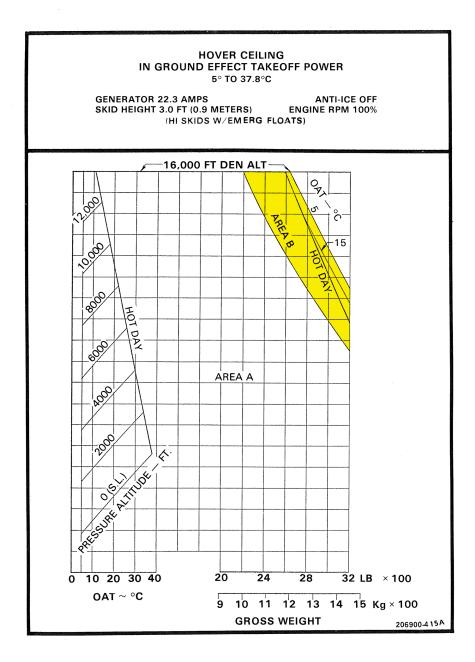


HOVERING CEILING - FLOATS STOWED

Out-of-ground-effect hovering performance is the same as basic helicopter. In-ground-effect hovering performance is shown on the following charts:

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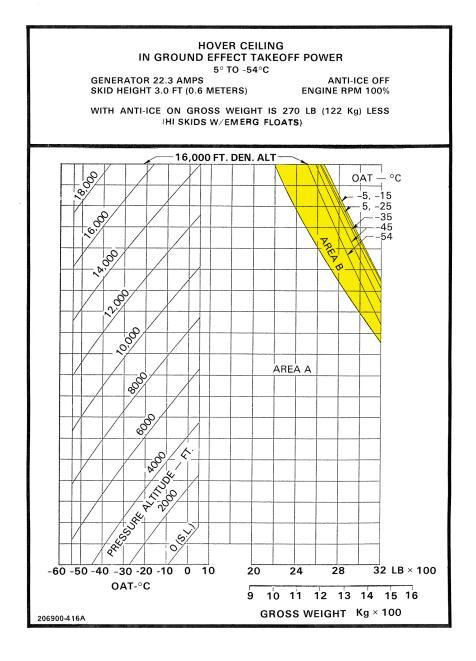


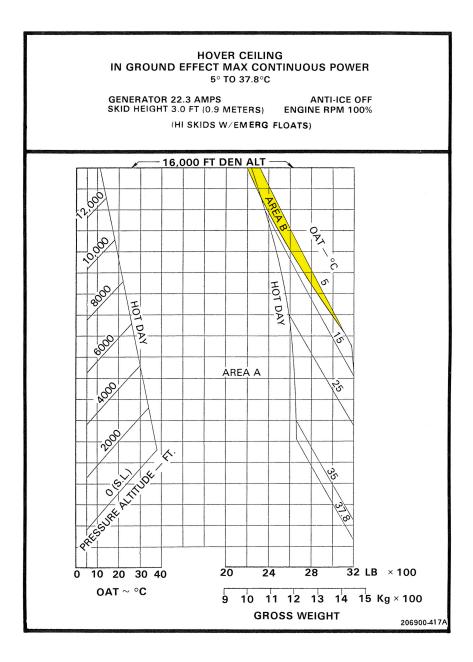
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