

**CESSNA AIRCRAFT COMPANY
MODEL 300 SERIES
CONTINUED AIRWORTHINESS PROGRAM**

TITLE **Wing Fuselage Attach Fittings Inspection**

MODEL	EFFECTIVITY	INSPECTION COMPLIANCE
336	336-0001 Thru 336-0195	INITIAL 12,000 HRS REPEAT 3000 HRS Thereafter
337 Thru T337H-SP	337-0002 Thru 33701955	
T337G Thru P337H	337-0002 Thru 33701955	
F337E/FT337E	F3370001 Thru F33700055	
FT337GP/FT337HP	FP33700001 Thru FP33700023	

PURPOSE

To eddy current inspect primary load path of forward and aft spar fitting.
There is one known history of cracking, except for rear spar web referred to in CAP Item 57-10-03.

INSPECTION INSTRUCTIONS

1. Inspect for possible cracking from wing attachment holes outward. (Refer to step 4.)
2. Possible galling or deformation of wing attachment holes. (Refer to step 4.)
3. Check wing attach bolts for looseness, If bolts are loose, remove wing attach bolts and inspect for cracks. (Refer to step 4.)
4. Eddy Current Inspect, per instructions in Section 4-0, "Basic Information."

ACCESS/LOCATION

Remove wing-to-fuselage fairing strips.

DETECTABLE CRACK SIZE

No Cracks Allowed

INSPECTION PROCEDURE

Eddy Current

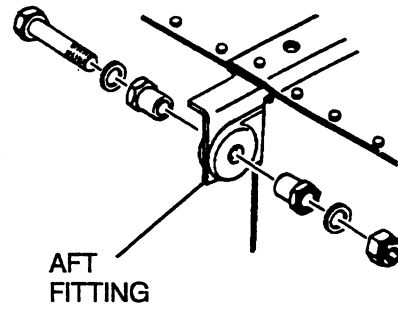
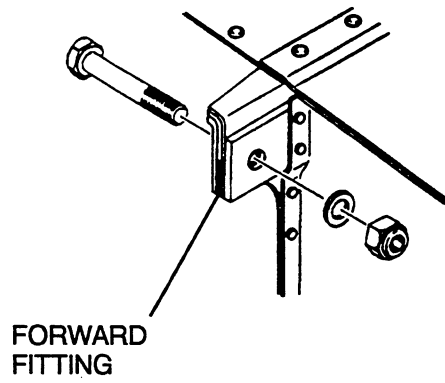
REPAIR/MODIFICATION

Replace cracked parts.

COMMENTS

Ref: ME74-4
Contact Cessna Aircraft Company for instructions if oversize holes are found.

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Wing Fuselage Attach Fitting
Figure 3-22

05221004
05221003