

**CESSNA AIRCRAFT COMPANY
MODEL 300 SERIES
CONTINUED AIRWORTHINESS PROGRAM**

TITLE **Wing Rear Spar Web**

MODEL

337 Thru T337H-SP
T337G Thru P337H
F337E/FT337E
FT337GP/FT337HP

EFFECTIVITY

337-0002 Thru 33701955
P3370001 Thru P3370356
F3370001 Thru F33700055
FP33700001 Thru FP33700023

INSPECTION COMPLIANCE

INITIAL 12,000 HRS

REPEAT 3000 HRS
Thereafter

PURPOSE

To ensure integrity of the wing spar web.

INSPECTION INSTRUCTIONS

1. Inspect web under wing rear spar fuselage attach fitting, paying particular attention to web inboard edge.

ACCESS/LOCATION

Remove wing-to-fuselage fairing strip.
Lower flaps and remove access cover.
Location: Web - Wing rear spar fuselage attach fitting area.

DETECTABLE CRACK SIZE

N/A

INSPECTION PROCEDURE

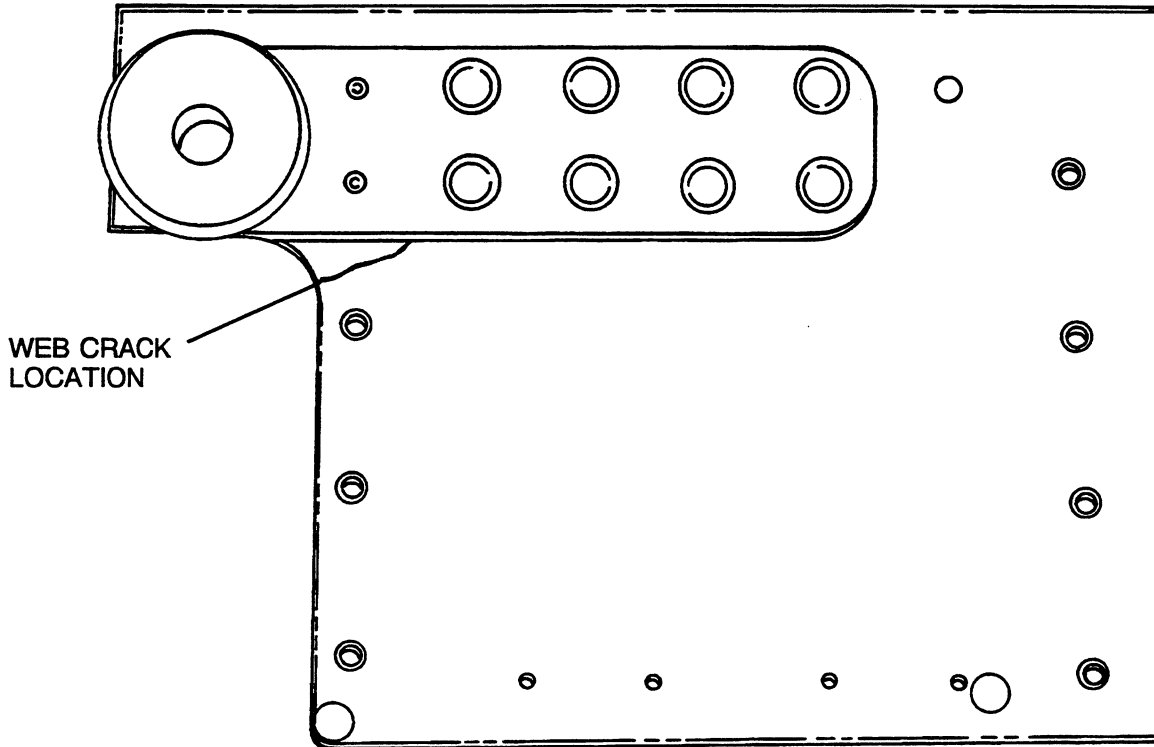
Visual

REPAIR/MODIFICATION

Report any evidence of cracking to Cessna Aircraft Company.

COMMENTS

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Rear Wing Spar Web - Inboard
Figure 3-23