

**CESSNA AIRCRAFT COMPANY  
MODEL 300 SERIES  
CONTINUED AIRWORTHINESS PROGRAM**

**TITLE**            **Wing Front Spar Inspection**

<b>MODEL</b>	<b>EFFECTIVITY</b>	<b>INSPECTION COMPLIANCE</b>
336	633, 363, 336-0001 Thru 336-0195	<b>INITIAL</b> 10,000 HRS *  <b>REPEAT</b> 5000 HRS * Thereafter
337 Thru T337H-SP	337-0002 Thru 33701955	
T337G Thru P337H	P3370001 Thru P3370356	
F337E/FT337E	F3370001 Thru F33700055	
FT337GP/FT337HP	FP33700001 Thru FP33700023	

**PURPOSE**

To check for cracks in front wing lower cap fittings.

**INSPECTION INSTRUCTIONS**

1. Visually inspect strut tube for cracks.
2. Eddy Current Inspect, per instructions in Section 4-0, "Basic Information."

**ACCESS/LOCATION**

**DETECTABLE CRACK SIZE**

Wing Root

No Cracks Allowed

**INSPECTION PROCEDURE**

Eddy Current and Visual

**REPAIR/MODIFICATION**

Replace both wing in accordance with applicable Service Manual, if cracks were found.

**COMMENTS**

Ref: ME72-26 Front Wing Spar Inspection December 12, 1972  
 Ref: ME76-3 Front Wing Spar Inspection January 9, 1976  
 Ref: ME78-2 Front and Rear Spar Inspection February 13, 1978  
 Report wing strut cracks to Cessna Aircraft Company.

\* Should any history of low level overland survey or patrol exist, then the initial inspection compliance is 3000-hours with repeat at 300-hours intervals thereafter.